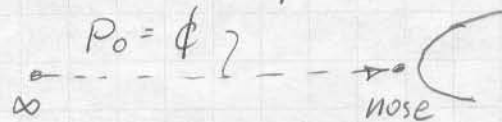


a)  $a_{\infty} = \sqrt{\gamma P_{\infty} / \rho_{\infty}}$  ,  $M_{\infty} \equiv V_{\infty} / a_{\infty} = \frac{V_{\infty} \sqrt{\rho_{\infty} / \gamma P_{\infty}}}{1}$

b) Assuming no bow shock (subsonic flight),  $P_{0\text{nose}} = P_{0\infty} \equiv P_0$



$P_{0\text{nose}} = P_{0\infty} = P_{\infty} \left[ 1 + \frac{\gamma-1}{2} M_{\infty}^2 \right]^{\frac{\gamma}{\gamma-1}}$  ( $P_0$  definition)

Using  $M_{\infty}$  result from a) ...

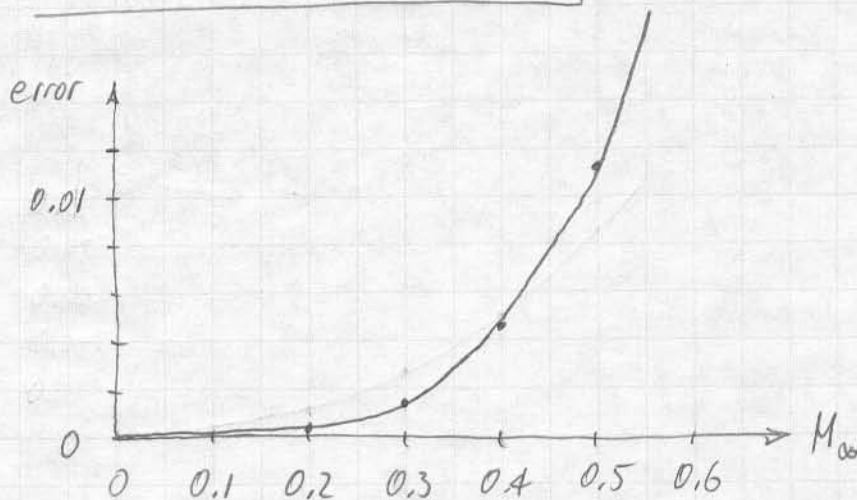
i)  $P_{0\text{nose}} = P_{\infty} \left[ 1 + \frac{\gamma-1}{2} \frac{\rho_{\infty} V_{\infty}^2}{\gamma P_{\infty}} \right]^{\frac{\gamma}{\gamma-1}}$

ii) Using Bernoulli (incorrectly):  $P_0 = P_{\infty} + \frac{1}{2} \rho_{\infty} V_{\infty}^2$

or  $P_0 = P_{\infty} \left[ 1 + \frac{\gamma}{2} \frac{\rho_{\infty} V_{\infty}^2}{\gamma P_{\infty}} \right] = P_{\infty} \left[ 1 + \frac{\gamma}{2} M_{\infty}^2 \right]$

error =  $\left( \frac{P_0}{P_{\infty}} \right)_{i)} - \left( \frac{P_0}{P_{\infty}} \right)_{ii)} = \left[ 1 + \frac{\gamma-1}{2} M_{\infty}^2 \right]^{\frac{\gamma}{\gamma-1}} - \left[ 1 + \frac{\gamma}{2} M_{\infty}^2 \right]$

$M_{\infty}$	error
0	0
0.1	0.0000175
0.2	0.00028
0.3	0.00143
0.4	0.00455
0.5	0.0112
0.6	0.0235



$M_{\infty} < 0.27$  gives  $\approx 0.1\%$  accuracy, seems pretty safe

$M_{\infty} < 0.47$  gives  $\approx 1\%$  accuracy, starts to get significant